

May 2004 CM



For a single plate about its center

$$\begin{aligned}
 I_z &= \int_{-L/2}^{L/2} \int_{-L/2}^{L/2} \rho_M (x^2 + y^2) dx dy \\
 &= 4 \int_0^{L/2} \int_0^{L/2} \rho_M (x^2 + y^2) dx dy \\
 &= 4 \rho_M \int_0^{L/2} \left. \frac{x^3}{3} + x y^2 \right|_{x=0}^{x=L/2} dy \\
 &= 4 \rho_M \int_0^{L/2} \left(\frac{(L/2)^3}{3} + \left(\frac{L}{2}\right) y^2 \right) dy \\
 &= 4 \rho_M \left(y \frac{(L/2)^3}{3} + \left(\frac{L}{2}\right) \frac{y^3}{3} \right) \Big|_{y=0}^{y=L/2} \\
 &= 4 \rho_M \left(\frac{2}{3} \left(\frac{L}{2}\right)^4 \right) \\
 &= 4 \frac{M}{L^2} \cdot \frac{1}{3} \frac{L^4}{8}
 \end{aligned}$$

$$I_z = \frac{1}{6} mL^2$$

$$\begin{aligned}
 I_x = I_y &= \int_{-L/2}^{L/2} \int_{-L/2}^{L/2} \rho_M x^2 dx dy \\
 &= 4 \rho_M \int_0^{L/2} \int_0^{L/2} x^2 dx dy \\
 &= 4 \rho_M \left(\frac{L}{2}\right) \frac{(L/2)^3}{3} \\
 &= 4 \frac{M}{L^2} \frac{1}{3} \frac{L^4}{16}
 \end{aligned}$$

$$I_x = I_y = \frac{1}{12} mL^2$$

By the parallel axis theorem, about point A:

$$I_x = \frac{1}{12} mL^2$$

$$I_y = \frac{1}{12} mL^2 + \frac{1}{4} md^2$$

$$I_z = \frac{1}{6} mL^2 + \frac{1}{4} md^2$$

Since the plates are ^{identical and} equidistant from A we have for both:

$$I_1 = \frac{1}{3} mL^2 + \frac{1}{2} md^2$$

$$I_2 = \frac{1}{6} mL^2 + \frac{1}{2} md^2$$

$$I_3 = \frac{1}{6} mL^2$$

$$b. \vec{L} = I_1 \omega_1 \hat{z} + I_2 \omega_2 \hat{y} + I_3 \omega_3 \hat{x}$$

$$\frac{d\vec{L}}{dt} \Big|_{lab} = \frac{d\vec{L}}{dt} \Big|_{rot} + \vec{\omega} \times \vec{L}$$

$$\begin{aligned}
 \vec{\tau} &= I_1 \dot{\omega}_1 \hat{z} + I_2 \dot{\omega}_2 \hat{y} + I_3 \dot{\omega}_3 \hat{x} \\
 &\quad + I_2 \omega_2 \omega_3 \hat{z} - I_1 \omega_1 \omega_3 \hat{y} - I_3 \omega_2 \omega_3 \hat{z} + I_1 \omega_1 \omega_2 \hat{x} \\
 &\quad + I_3 \omega_1 \omega_3 \hat{y} - I_2 \omega_1 \omega_2 \hat{x}
 \end{aligned}$$

Or in component form:

$$\tau_z = I_1 \dot{\omega}_1 + (I_2 - I_3) \omega_2 \omega_3 \quad (1)$$

$$\tau_y = I_2 \dot{\omega}_2 + (I_3 - I_1) \omega_1 \omega_3 \quad (2)$$

$$\tau_x = I_3 \dot{\omega}_3 + (I_1 - I_2) \omega_1 \omega_2 \quad (3)$$

$\vec{\tau} = 0$, initially only ω_2 is non-zero.

Immediately after the asteroid hits, ω_1 and ω_3 are initially

small but non-zero. Let: $\omega_1 = \omega_{10} e^{st}$ (s complex)

$$\omega_3 = \omega_{30} e^{st}$$

In (2) $\omega_1, \omega_3 \sim \epsilon^2$ and can be neglected

thus $I_2 \dot{\omega}_2 \approx 0$ and ω_2 can be taken as constant

$$\therefore (1) \Rightarrow s I_1 \omega_1 + (I_2 - I_3) \omega_2 \omega_3 = 0$$

$$(3) \Rightarrow s I_3 \omega_3 + (I_1 - I_2) \omega_1 \omega_2 = 0$$

$$\omega_3 = \frac{1}{s I_3} (I_2 - I_1) \omega_2 \omega_1$$

$$s I_1 \omega_1 + (I_2 - I_3) \omega_2 \cdot \frac{1}{s I_3} (I_2 - I_1) \omega_2 \omega_1 = 0$$

$$s^2 I_1 I_3 + \omega_2^2 (I_2 - I_3)(I_2 - I_1) = 0$$

$$s^2 = -\omega_2^2 \frac{(I_2 - I_3)(I_2 - I_1)}{I_1 I_3}$$

Since $I_1 > I_2 > I_3$, $s^2 > 0$

Thus ω_1 and ω_3 represent a superposition of exponentially growing and exponentially decaying solutions.

The growing solution dominates and the motion is perturbed strongly.

$$\tau = \frac{1}{s} = \omega_2 \sqrt{\frac{I_1 I_3}{(I_2 - I_3)(I_1 - I_2)}}$$

Solving for ω_2 :

In the rotating frame:

$$\vec{F} = -G \frac{m^2}{(d+L)^2} \quad (\text{Force on the center of one panel is due to}$$

only the other panel by symmetry, also approx $L \ll d$)

$$\vec{a}|_{\text{lab}} = \vec{a}|_{\text{rot}} + \vec{\omega}_2 \times \vec{V}$$

$$-\frac{g}{6} = -G \frac{m}{(d+L)^2} - \omega_2 \cdot \omega_2 \cdot \frac{1}{2} (d+L)$$

$$\frac{g}{3(d+L)} - 2G \frac{m}{(d+L)^3} = \omega_2^2$$

$$\omega_2 = \sqrt{\frac{g}{3(d+L)} - 2G \frac{m}{(d+L)^3}}$$